

## Progress Report of the

## ATA / IATA / SAE Commercial Aircraft Composite Repair Committee (CACRC)

Presented at : CACRC Main Committee Meeting

Prepared by Carlos Blohm Issue : November, 2007 History: The CACRC is an airline maintenance committee, formed in 1991 as a combination of ATA, IATA and SAE committees, with a common charter.

## Charter:

"To develop and improve maintenance, inspection and repair of commercial aircraft composite structure and components"

Ultimate goal:

- Lower maintenance cost of existing composite structures, via standardization among OEMs and airlines.
- ♦ Minimize life cycle cost of future designs.

## Philosophy:

- Create/establish standard solutions to the most common airline problems.
- Use task groups of industry experts to discuss the existing solutions and write specifications.

## Scope:

- Address areas that impact airlines most frequently.
- Consensus documents Disapproving votes to be resolved
  - 1. Where consensus is not reached, all options are listed
  - 2. Not a research organization best practices are selected

## **Organization of CACRC**

### **Entities:**

- Airline Operators
- Aircraft Original Equipment Manufacturers
- FAA, EASA / JAA, repair stations, vendors, material suppliers, training institutes, academia, any other interested parties
- SAE as secretariat and publisher of standards

## <u>Meetings:</u>

• Main Committee: twice per year

(alternating between Europe / N. America)

- Executive Committee: before and after Main Committee
- Task Groups: 2 or 3 working meetings per year

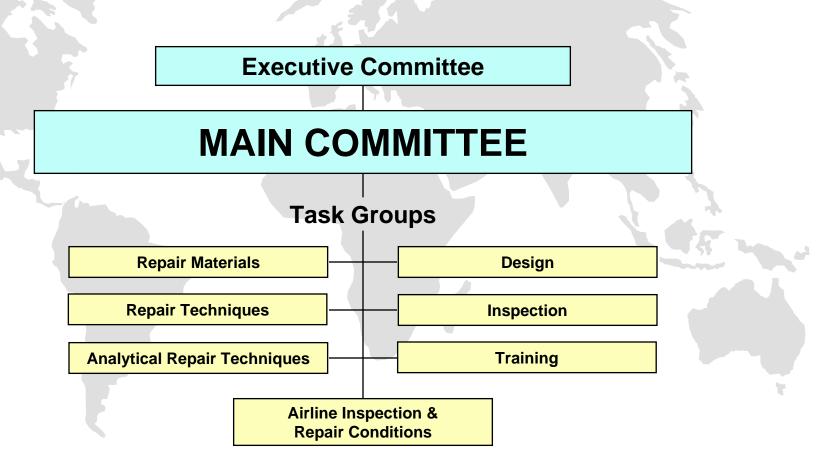
## Membership of CACRC

**USER MEMBER**: A commercial transport air operator, a manufacturer of an aerospace vehicle subsystem or part, or a government organization. Only User Members Vote on documents.

- **SUPPLIER MEMBER**: Producers of the various materials and processes are an integral part of the AMS operation. Supplier members have no formal vote but can comment on documents.
- **LIAISON MEMBER**: Liaisons relay information to and from parallel activities of other committees and organizations. Liaisons have no formal vote but can comment on documents.

 CONSULTANT MEMBER: A person having specific technical knowledge. Consultants have no formal vote but can comment on documents.
 MAILING LIST: A person that only receives the minutes, agendas, and announcements. Does not receive balloted documents.

## **Structure of CACRC**



## **Task Group Summary**

## <u>Repair Materials Task Group</u>

Chairperson: Dr. Ana Rodriguez – Airbus

<u>Charter</u>: To develop "Common Repair Material Specifications" and support the qualification process.

## Repair Techniques Task Group

Chairperson: Francois Museux - Airbus
Charter: To develop "Standardized Repair Techniques"

## Analytical Repair Techniques Task Group

Chairperson: Tim Harris – Boeing, Ray Kaiser - Northwest

 <u>Charter</u>: To develop a guide, for commercial aircraft composite engineers, containing generally accepted analytical techniques to enable a conservative analysis for repair justification, as NAA-approvable data.

## Task Group Summary (cont.)

### Design Task Group

Chairperson: Eric Chesmar – United Airlines

<u>Charter</u>: Task 1: To improve composite design through operator feedback.

Task 2: To develop a "Maintenance Life Cycle Cost Model for Commercial Aircraft Composite"

## Inspection Task Group

Chairperson: John Hewitt - Airbus, Dennis Roach - Sandia Laboratories

<u>Charter</u>: Task 1: To develop a "Standard Guideline for Composite inspection".

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Task 2: Steering Committee for "Composite Reference Standards" program at FAA and Sandia Labs.

## Task Group Summary (cont.)

## Training Task Group

Chairperson: Joe Hafenrichter – Boeing <u>Charter</u>: To develop guidelines for composite training.

## Airline Inspection& Repair Conditions Task Group

Chairperson: William F. Cole and John Player – United Airlnes, Oksana Bardygula - FedEx

<u>Charter</u>: Task 1: To define time, equipment, training and materials available for normal airline operation, so OEMs can design inspection and repair schemes for composite structure, with minimum impact. Task 2: Prioritize Parts for enlarged Allowable Damage Limits and Repairable Damage Limits.

## **Progress Status**

**Published documents** 

- 12 SAE / Aerospace Materials Specifications (AMS)
- ♦ 6 SAE / Aerospace Information Reports (AIR)
- 10 SAE / Aerospace Recommended Procedures (ARP)
- I SAE / Aerospace Engineering Report (AE)

Documents have been implemented in OEM Manuals and Procedures!! (see document implementation status)

7 further documents in development or in revision cycle



## **Commercial Aircraft Composite Repair Committee**

Document sponsor	Document	Status	Implementation
Main Committee	AC 145-6 Repair Station for Composite and Bonded Aircraft Structure	issued	used as reference by: a) FAA inspectors auditing Repair Stations b) MRO establishing procedures and quality manuals
T/G Repair Materials	AMS 2980, /1, /2, /3, /4 Technical Specification"Carbon Fiber Fabric and Epoxy Resin Wet Lay-Up Repair material"	issued 01.Nov.96 revised	Qualification completed a) A.I.: incorporated in AIMS 08-01-00 b) Boeing : c) P&W : d) GE : e) Goodrich: Certification in 2007 / SRM incorporation after certification.
T/G Repair Materials	AMS 3970, /1, /2, /3, /4 Technical Specification"Carbon Fiber Fabric Repair Prepreg, 125Mdc (250Mdf) Vacuum curing"	issued 01.Dec.99 revision balloted	Material selection for qualification under progress
T/G Repair Materials	AMS 2960, /1, /5 Glass Fabric with Epoxy Resin Wet Lay- Up Repair Material	in progress	7
T/G Repair Materials	AMS 2950, /1 Paste Adhesive and Core Restoration	balloted	
T/G Repair Materials	AMS XXXX Glass Prepreg for structural application	in progress	

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Document implementation status			
Document sponsor	Document	Status	Implementation
T/G Repair Techniques	ARP 4916 Masking and Cleaning of Epoxy and Polyester Matrix Thermosetting Composite Materials.	issued 01.Mar.97 reaffirmed 28.Jun.06	<ul> <li>a) A.I. : Methods reflecting AI process included in SRM 51-77</li> <li>b) Boeing :</li> <li>c) P&amp;W :</li> <li>d) GE : implemented completely in Standard Practices Manual 70-46-01</li> <li>e) Goodrich: CSSP 540258 Pre-Preg repair (adapted)</li> </ul>
T/G Repair Techniques	ARP 4977 Drying of Thermosetting Composite Materials.	issued 01.Aug.96 reaffirmed 28.Jun.06	<ul> <li>a) A.I. : implemented</li> <li>b) Boeing : commitment to reference in SRM in near future</li> <li>c) P&amp;W : implemented in Standard Practices Manual 70-46-02</li> <li>d) GE : used as a resource in the answer of customer questions</li> <li>e) Goodrich: Standard Process CSSP 540252 Moisture Removal / CFM, A1/A5, PW SRM 54</li> </ul>
T/G Repair Techniques	ARP 5256 Mixing Resins, Adhesives and Potting Compounds	issued 01.Mar.97 reaffirmed 28.Jun.06	<ul> <li>a) A.I. : Used in training course</li> <li>b) Boeing :</li> <li>c) P&amp;W : implemented in 1997</li> <li>d) GE : used as a resource in the answer of customer questions</li> <li>e) Goodrich: Standard Process CSSP 540222 / V2500 A1/A5 SRM 54-02 (adapted)</li> </ul>
T/G Repair Techniques	ARP 4991, A Core Restoration of Thermosetting Composite Materials	issued 01.Dec.98 revised 27.Feb.06	<ul> <li>a) A.I. : existing technique in SRM 51-77 very close to ARP</li> <li>b) Boeing :</li> <li>c) P&amp;W : implemented in1998</li> <li>d) GE :</li> <li>e) BF Goodrich: Standard Process 540254 / V2500 A1/A5 SRM 54 (adapted)</li> </ul>
T/G Repair Techniques	ARP 5143 Vacuum Bagging of Thermosetting Composite Repairs	issued 26.Jul.02	<ul> <li>a) A.I. : Methods reflecting AI process included in SRM 51-77</li> <li>b) Boeing :</li> <li>c) P&amp;W : implemented in 1998</li> <li>d) GE : implemented in Standard Practices Manual 70-46-03</li> <li>e) Goodrich: CSSP 540251 Vacuum Bag Process / V2500</li> <li>A1/A5 SRM 54-02 (adapted)</li> </ul>

Document sponsor	Document	Status	Implementation
T/G Repair Techniques	ARP 5144 Heat Application for Thermosetting Resin Curing	issued 01.Mar.00 reaffirmed 28.Jun.06	<ul> <li>a) A.I. : implemented in SRM 51-77 (2000)</li> <li>b) Boeing :</li> <li>c) P&amp;W : implemented in 1998</li> <li>d) GE : ARP is referenced directly in SPM 70-46-02 and 70-46-03</li> <li>e) Goodrich: Standard Process CSSP 540221 and 540222 (adapted)</li> </ul>
T/G Repair Techniques	ARP 5319 Impregnation of Dry Fabric and Ply Lay- Up	issued 19.Jul.02	<ul> <li>a) A.I. : vertical bleeding implemented in SRM 51-77 (2000)</li> <li>b) Boeing :</li> <li>c) P&amp;W :</li> <li>d) GE :</li> <li>e) Goodrich: Standard Process 540256 / V2500 A1/A5 SRM 54 (adapted)</li> </ul>
T/G Repair Techniques	AIR 5367 Machining of Epoxy and Polyester Matrix Thermosetting Composite Structures	in progress	a) A.I. : b) Boeing : c) P&W : d) GE : e) Goodrich:
T/G Repair Techniques	AIR 5431 Repair Tooling	Issued 29.Dec.04	a) A.I. : b) Boeing : c) P&W : d) GE : e) Goodrich:
T/G Repair Techniques	ARP 5701 Storage and Handling of Epoxy Thermosetting Composite Materials	in progress	a) A.I. : b) Boeing : c) P&W : d) GE : e) Goodrich:
T/G Repair Techniques	AIR 5702 Lay-up of Prepreg Composite Materials	in progress	a) A.I. : b) Boeing : c) P&W : d) GE : e) Goodrich:

<u>Document implementation status</u>				
Document sponsor	Document	Status	Implementation	
T/G Inspection	ARP 5089 Composite Repair NDT / NDI Handbook	issued 01.Nov.96	FAA: internal use Boeing: internal use A.I.: Internal use Goodrich: Internal use, CSSP 540258 Pre-Preg repair.	
T/G Inspection	ARP 5605 Solid Composite Laminate NDI Reference Standards	issued 01.Sep.01	a) A.I. : Internal use b) Boeing : c) P&W : d) GE : e) Goodrich:	
T/G Inspection	ARP 5606 Composite Honeycomb NDI Reference Standards	issued 01.Sep.01	<ul> <li>a) A.I. : Range of possible applications under review</li> <li>b) Boeing :</li> <li>c) P&amp;W :</li> <li>d) GE :</li> <li>e) Goodrich:</li> </ul>	
T/G Design	AE-27 Design of Durable, Repairable and Maintainable Aircraft Composites	issued 11.Jul.97	<ul> <li>a) Presentation to European designers Nov.97.</li> <li>b) Presentation to BF Goodrich Aerospace designers Feb.98.</li> <li>c) Presentation to Boeing designers May 98.</li> <li>d) Presentation at SAMPE conference</li> <li>e) Goodrich: implemented in Design &amp; Best Practice Nacelle manuals.</li> <li>f) P&amp;W: implemented in Nacelle Design Handbook.</li> <li>g) Boeing: implemented in Design Handbook.</li> <li>h) A.I.: presented to partners and implemented in Design Hbk.</li> <li>i) Presentation at EADS Stade Sep 01.</li> </ul>	
T/G Design	AIR 5416 Maintenance Life Cycle Cost Model	in balloting		

Document sponsor	Document	Status	Implementation	
T/G Training	AIR 4938, A Composite and Bonded Structure Technician / Specialist: Training Doc.	issued 01.Sep.96 revision balloted	Flight Safety: uses as basis (100%) A.I.: Reduced content used, to match course length	
T/G Training	AIR 5278 Composite and Bonded Structure Engineers: Training Doc.	issued 01.Mar.99 reaffirmed 26.Jan.06	Flight Safety: uses as basis (100%) A.I.: Reduced content used, to match course length	
T/G Training	AIR 5279 Composite and Bonded Structure Inspector: Training Doc.	issued 01.Mar.99 reaffirmed 26.Jan.06	Flight Safety: uses as basis (100%) A.I.: Reduced content used, to match course length	
T/G Training	AIR 4844, A, B, C Composites and Metal Bonding Glossary	issued 01.Mar.97 3 <sup>rd</sup> revision in progress	<ul> <li>a) A.I. : Internal use</li> <li>b) Boeing :</li> <li>c) P&amp;W :</li> <li>d) GE : used as a resource in house</li> <li>e) Goodrich: Uses as reference in-house and with customers</li> </ul>	
T/G Training	Composite Structures Awareness, Video	available		
T/G Training	Proper Handling of Composite Parts, Video	available		
T/G Training	AIR 5719 Teaching Points for an Awareness Class on "Critical Issues in Composite Maintenance and Repair"	balloted		

Document sponsor	Document	Status	Implementation
T/G Airline Inspection &	Structural Repair Manual Limitations in	issued	
Repair Conditions	Commercial Airline Maintenance,	30.Oct.98	
	Report		
			<i>.</i>
T/G Analytical Repair	AIR 5946	in Progress	
Techniques	Design and Application of Composite Repairs for Thermosetting Composites		

## **Other Successes for CACRC to date:**

Forum to act as the source for industry to obtain airline input / feedback, such as :

- Comments on Advisory Circular Number 145-6 "Repair Station for Composite and Bonded Aircraft Structures"
- Sandia NDT Program Steering group.
- ♦ FAR Revisions
- Airline contacts for design reviews.
- Participation at FAA Composite Workshops.

## <u>But,</u>

# Most of the issues that existed 20 years ago still exist!

## Materials

- > Numerous types, styles and procurement specifications.
- Limited and/or untimely availability.
- Minimum purchase amounts and cost associated with the expiration of shelf life.
- > No standard means of determining substitutability.

## Repair Techniques

- > Limited options, Component specific, OEM driven.
- Inability to standardize materials and repair processing due to unknown effects on design properties.
- > Limited independent analysis capabilities.
- Denied access to design loads and material allowables necessitating innovative repair solutions.

# Most of the issues that existed 20 years ago still exist! (cont.)

#### Repair Access

- > Repairability in design.
- Self inflicted damage associated with obtaining back side access to damaged areas.
- Cost associated with lack of access.

## Tooling and Equipment

- > Inability to obtain OEM loft data.
- Concern over exact replication of OEM tooling.
- > Compatibility of tooling material selection.
- Consideration of thermal coefficient of expansion.
- > Repair spring back vs. manufacturing spring back.

# Most of the issues that existed 20 years ago still exist! (cont.)

## Approval Legalities

- > All process and/or material deviations require formal approval.
- > A DER is the main means of obtaining FAA repair approval.
- Repair processing must be approved by the customers engineering department.
- > Lack of trust in bonded repairs.

### Training

> Dedicated training and licensing is still not mandatory.

## **Summary**

- Operator input remains absolute necessity. The attendance of airline / MRO personnel has to be encouraged.
- CACRC proved to be effective to improve maintainability of composite structures, but focus has to be extended to primary structures.
- CACRC proves to be worthwhile for operators, OEM's and material supplier, but results must be implemented faster
- CACRC is breaking new ground in industry and regulatory agencies cooperation.

## Next meetings

## **European Spring meeting:**

♦ tbd

## North American Fall meeting: tbd

Main committee detailed meeting information can be found on the SAE web page: <u>http://works.sae.org</u>

## You are invited to join CACRC

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